Lecture 1 Classical Feedback Control

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The lifting force :

KVL :

$$
f(t) = K \frac{i^2(t)}{y(t)}
$$

$$
Ri(t) + L\frac{di(t)}{dt} = v(t)
$$

Newton's Law:

$$
M\frac{d^2y(t)}{dt^2} = -K\frac{i^2(t)}{y(t)} + Mg
$$

The differential equation model

$$
Ri(t) + L\frac{di(t)}{dt} = v(t)
$$

$$
M\frac{d^2y(t)}{dt^2} = -K\frac{i^2(t)}{y(t)} + Mg
$$

Selecting $x_1(t) = i(t)$, $x_2(t) = y(t)$, $x_3(t) = \dot{y}(t)$ and $u(t) = v(t)$, the state-space model is

$$
\begin{bmatrix}\n\dot{x}_1(t) \\
\dot{x}_2(t) \\
\dot{x}_3(t)\n\end{bmatrix} = \begin{bmatrix}\n\frac{u(t) - Rx_1(t)}{L} \\
x_3(t) \\
-x_3^2(t) \\
-\frac{Kx_1^2(t)}{Mx_2(t)} + g\n\end{bmatrix}
$$
\n
$$
y = x_2(t)
$$

Linearize the system around an operating point with $y(t) = y_0$.

Linearization

Definition

A triple of constant vectors $\begin{bmatrix} u_0 & x_0 & y_0 \end{bmatrix} \in \mathbb{R}^n \times \mathbb{R}$ is said to be an *operating point* of the system if

> $f(x_0, u_0) = 0$ $g(x_0, u_0) = y_0$

$$
\frac{u_0 - Rx_{10}}{L} = 0, \qquad x_{30} = 0
$$

$$
-\frac{Kx_{10}^2}{Mx_{20}} + g = 0, \qquad x_{20} = y_0
$$

This gives

$$
u_0 = R\sqrt{\frac{Mgy_0}{K}},
$$
 $\begin{bmatrix} x_{10} \\ x_{20} \\ x_{30} \end{bmatrix} = \begin{bmatrix} \sqrt{\frac{Mgy_0}{K}} \\ y_0 \\ 0 \end{bmatrix},$ $y_0 = y_0$

Magnetic-Ball Suspension System Linearization

The physical meaning of an operating point is that if the system has initial condition *x*⁰ and a constant input u_0 is applied, then the state and output will stay at constant values x_0 and *y*0, respectively, for all time, i.e.,

$$
u(t) = u_0
$$
, $x(0) = x_0 \Rightarrow x(t) = x_0$, $y(t) = y_0$.

Since *f* and *g* are sufficiently smooth, we can conclude that

$$
u(t) - u_0, x(0) - x_0 \text{ are small } \Rightarrow x(t) - x_0, y(t) - y_0 \text{ are small.}
$$

Denote $\tilde{u}(t) = u(t) - u_0$, $\tilde{x}(t) = x(t) - x_0$ and $\tilde{y}(t) = y(t) - y_0$.

$$
\dot{\tilde{x}}(t) = A\tilde{x}(t) + b\tilde{u}(t), \qquad \tilde{y}(t) = c\tilde{x}(t) + d\tilde{u}(t)
$$

$$
A = \frac{\partial f}{\partial x}\Big|_{x=x_0, u=u_0}, B = \frac{\partial f}{\partial u}\Big|_{x=x_0, u=u_0}
$$

$$
C = \frac{\partial g}{\partial x}\Big|_{x=x_0, u=u_0}, D = \frac{\partial g}{\partial u}\Big|_{x=x_0, u=u_0}
$$

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Linearization

$$
\tilde{u}(t) = u(t) - u_0 = u(t) - R\sqrt{\frac{Mgy_0}{K}}
$$

$$
\tilde{x}(t) = x(t) - x_0 = \begin{bmatrix} x_1(t) - \sqrt{\frac{Mgy_0}{K}} \\ x_2(t) - y_0 \\ x_3(t) \end{bmatrix}, \ \tilde{y}(t) = y(t) - y_0
$$

The linearized model of the deviation variables is

$$
\begin{aligned} \dot{\bar{x}}(t) &= \begin{bmatrix} -\frac{R}{L} & 0 & 0 \\ 0 & 0 & 1 \\ -2\sqrt{\frac{gK}{My_0}} & \frac{g}{y_0} & 0 \end{bmatrix} \tilde{x}(t) + \begin{bmatrix} \frac{1}{L} \\ 0 \\ 0 \end{bmatrix} \tilde{u}(t) \\ \tilde{y}(t) &= \begin{bmatrix} 0 & 1 & 0 \end{bmatrix} \tilde{x}(t) + \begin{bmatrix} 0 \end{bmatrix} \tilde{u}(t) \end{aligned}
$$

Magnetic-Ball Suspension System Transfer Function

Transfer Function

$$
G(s) = \frac{Y(s)}{U(s)} = C(sI - A)^{-1}B + D
$$

$$
G(s) = \begin{bmatrix} 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} s + \frac{R}{L} & 0 & 0 \\ 0 & s & -1 \\ -2\sqrt{\frac{gK}{My_0}} & -\frac{g}{y_0} & s \end{bmatrix}^{-1} \begin{bmatrix} \frac{1}{L} \\ 0 \\ 0 \end{bmatrix}
$$

$$
= \frac{-2\sqrt{\frac{gK}{My_0}} \frac{1}{L}}{\left(s + \frac{R}{L}\right)\left(s^2 - \frac{g}{y_0}\right)} = \frac{-2\sqrt{gy_0K}}{\sqrt{M}\left(Ls + R\right)\left(y_0s^2 - g\right)}
$$

One degree-of-freedom controller

Objectives:

- \blacktriangleright Closed-loop stability
- \blacktriangleright Reference tracking
- \blacktriangleright Disturbance rejection
- \blacktriangleright Noise response

One degree-of-freedom controller

The input to the plant is

 $u = K(s)(r - y - n)$

The control error *e* is defined as

e = *y − r*

Closed-loop transfer functions

The plant model : $y = G(s)u + G_d(s)d$

For one-degree-of-freedom controller

$$
y = GK(r - y - n) + G_d d \hspace{0.05cm}
$$

or

$$
(I+GK)y = GKr + G_d d - GKn
$$

hence

$$
y = \underbrace{(I+GK)^{-1}GK}_{T}r + \underbrace{(I+GK)^{-1}G_d d}_{S} - \underbrace{(I+GK)^{-1}GK}_{T}n
$$

$$
e = y - r = -Sr + SG_d d - Tn
$$

u = *KSr − KSGdd − KSn*

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Closed-loop transfer functions

Loop transfer function :

$$
L(s) = G(s)K(s)
$$

Sensitivity function :

$$
S(s) = (I + GK)^{-1} = (I + L)^{-1}
$$

Complementary sensitivity function :

$$
T(s) = (I + GK)^{-1} GK = (I + L)^{-1}L
$$

S is the closed-loop transfer function from the output disturbances to the outputs. *T* is the closed-loop transfer function from the reference signals to the outputs.

$$
S + T = (I + L)^{-1} + (I + L)^{-1}L = I
$$

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Feedback Control Sensitivity

Sensitivity to plant gain changes

$$
S(s) = \frac{\text{relative closed-loop response change}}{\text{relative open-loop response change}}
$$

$$
= \frac{dT(s)/T(s)}{dG(s)/G(s)}
$$

An change of *α* percent in the open-loop plant DC gain gives a change of *αS*(0) percent in the closed-loop DC gain.

Performance requirements

Feedback Control Why feedback?

"Perfect" control can be obtained, even without feedback $(K = 0)$ by

 $K_r(s) = G^{-1}(s);$ $K_d(s) = G^{-1}(s)G_d(s)$

Let $u = K_r r - K_d d$ and we get

$$
y = G(G^{-1}r - G^{-1}G_d d) + G_d d = r
$$

Feedback Control Why feedback?

The fundamental reasons for using feedback control are therefore the presence of

- \triangleright Signal uncertainty unknown disturbance (d)
- \blacktriangleright Model uncertainty (Δ)
- An unstable plant unstable plants can only be stabilized by feedback.

Closed-loop Stability

Inverse response process

Consider a system $G(s) = \frac{3(-2s+1)}{(5s+1)(10s+1)}$. If we close the feedback loop using a proportional gain *K*, the step responses of the closed-loop system *y*(*t*) are shown below:

- \triangleright Check the closed-loop poles That is the roots of $1 + L(s) = 1 + KG(s)$. The system is stable if and only if all roots are in left half plane (LHP).
- **►** Use Bode' stability condition: Stability \Leftrightarrow $|L(j\omega_{180})|$ < 1.
- \blacktriangleright Use Nyquist' stability criterion.

Bode plot margins

Nyquist plot margins

Closed-loop performance

$$
G(s) = \frac{5e^{-0.1s}}{(s+1)(0.1s+1)} \quad K(s) = \frac{0.5s+1}{s}
$$

Closed-loop performance Bandwidth

- \blacktriangleright (Closed-loop) bandwidth $ω_B$: Frequency at which $|S(jω)| = -3dB = 1/\sqrt{2}$
- **I** Crossover frequency ω_c : Frequency at which $|L(j\omega)| = 1$.
- \blacktriangleright if PM < 90^{*◦*} then $\omega_B < \omega_c < \omega_{BT}$ **Lecture 1 Classical Feedback Control** J **19/27** \triangleright **4 19/27** \tri

Loop shaping

- \blacktriangleright the classical loop-shaping approach is to shape the magnitude of the loop transfer function $L(s) = G(s)K(s)$ via the designed controller $K(s)$.
- \blacktriangleright Trade-offs in terms of L

$$
e = y - r = -\underbrace{(I + L)^{-1}}_{S} r + \underbrace{(I + L)^{-1}}_{S} G_d d - \underbrace{(I + L)^{-1}L}_{T} n
$$

- ▶ Performance, good disturbance rejection: needs large controller gains, i.e. *L* large.
- \triangleright Good command following, Stabilization of unstable plant: *L* large
- \blacktriangleright Reduce measurement noise, Nominal stability, Robust stability: L small.
- ▶ We need a large loop gain ($|L| > 1$) at low frequencies below crossover, and a small gain $(|L| < 1)$ at high frequencies above crossover.
	- **I** Define he gain crossover frequency, ω_c , where $|L(j\omega_c)| = 1$.
	- \blacktriangleright We desire a slope of 20 dB/dec (or 1 in log-log scale) around crossover, and large roll-off at high frequencies. The desired slope at lower frequencies depends on the nature of the disturbance or reference signal.
	- \blacktriangleright Define the system type that is the number of pure integrators in $L(s)$.

Loop shaping

Example

Consider
$$
G(s) = \frac{3(-2s+1)}{(5s+1)(10s+1)}
$$
 and $L(s) = 3K \frac{(-2s+1)}{s(2s+1)(0.33s+1)}$

$$
\frac{9}{2} \times \frac{10^{9}}{10^{2}}
$$

$$
= \frac{90}{10^{2}}
$$

$$
= \frac{90}{6} \times \frac{10^{2}}{10^{-2}}
$$

$$
= \frac{90}{10^{-2}}
$$

$$
= \frac{90}{10^{-2}}
$$

$$
= \frac{90}{10^{-2}}
$$

$$
= \frac{10^{6}}{10^{6}}
$$

Frequency [rad/s]

Loop shaping Example

The controller corresponding to the loop-shape is $K(s) = 0.05 \frac{(10s+1)(5s+1)}{s(2s+1)(0.33s+1)}$.

Disturbance rejection

Example

Using a controller $K_0(s) = \frac{1}{s}$ $10s + 1$ 200 $0.1s + 1$ $\frac{0.128 + 1}{0.01s + 1}$, which is for reference tracking objective we have

Disturbance rejection Example

Same problem but this time we concentrate on disturbance rejection. The controller is then $K_3(s) = 0.5 \frac{s+2}{s}$ *s* $0.05s + 1$ $\frac{0.005s + 1}{0.005s + 1}$. We have

- \blacktriangleright reference tracking is not good: large overshoot.
- \blacktriangleright disturbance rejection is good.
- to solve both objectives simultaneous, one can use a two-degree-of-freedom controller.

Two-degrees-of-freedom control

 \blacktriangleright We use the same feedback controller as in previous example.

Here $K_r = \frac{0.5s + 1}{0.65}$ $0.65s + 1$ 1 $\frac{1}{0.03s+1}$.

 \blacktriangleright K_r can be used to improve the tracking reference separately from the disturbance rejection

$$
T_{fb} = K_r \frac{G K_y}{1 + G K_y}, \qquad T_d = G_d \frac{1}{1 + C}
$$

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 $1 + GK_y$

Two-degrees-of-freedom control

 \blacktriangleright Finally, we are able to satisfy all requirement using a two-degree-of-freedom controller.

Reference

- 1 S. Skogestad and I. Postelthwaite, *Multivariable Feedback Control: Analysis and Design*, 2nd Edition, Wiley.
- 2 J. C. Doyle, B. A. Francis and A. R. Tannenbaum, *Feedback Control Theory*, McMillan, 1992.