

# Lecture 11: Linear Matrix Inequalities

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# Linear Matrix Inequalities

A linear matrix inequality (LMI) has the form

$$M(p) = M_0 + p_1 M_1 + \cdots + p_N M_N < 0$$

where  $M_0, M_1, \dots, M_N$  are given symmetric matrices,  $p = [p_1 \quad p_2 \quad \dots, p_N]^T$  is a column vector of real scalar variables.

- ▶ the matrix inequality  $M(p) < 0$  means that the left hand side is negative definite.
- ▶ An important property of LMIs is that the set of all solutions  $p$  is convex.
- ▶ LMIs can be used as constraints for the minimization problem

$$\min_p c^T p \text{ subject to } M(p) < 0$$

where the elements of the vector  $c$  in the linear cost function are weights on the individual decision variables.

# Linear Matrix Inequalities

- ▶ the convex problem can be solved by efficient, polynomial-time interior-point methods.
- ▶ Several LMI constraints can be combined into a single constraint of type.
- ▶ for example the constraint

$$M_1(p) < 0 \quad \text{and} \quad M_2(p) < 0$$

is equivalent to the single LMI constraint

$$\begin{bmatrix} M_1(p) & 0 \\ 0 & M_2(p) \end{bmatrix} < 0$$

# Pole Region Constraints

- ▶ the condition that the poles of a system are located within a given region in the complex plane can be formulated as an LMI constraint.
- ▶ the dynamic system  $\dot{x}(t) = Ax(t)$ . This system is stable if and only if the matrix  $A$  has all eigenvalues in the left half plane, which is true iff there exists a positive definite, symmetric matrix  $P$  that satisfies the Lyapunov inequality

$$PA^T + AP < 0$$

- ▶ This inequality is linear in the matrix variable  $P$ , and one can use LMI solvers to search for solutions.
- ▶ assume that  $A$  is a 2 by 2 matrix and write the symmetric matrix variable  $P$  as

$$P = \begin{bmatrix} p_1 & p_2 \\ p_2 & p_3 \end{bmatrix} = p_1 \begin{bmatrix} 1 & 0 \\ 0 & 0 \end{bmatrix} + p_2 \begin{bmatrix} 0 & 1 \\ 1 & 0 \end{bmatrix} + p_3 \begin{bmatrix} 0 & 0 \\ 0 & 1 \end{bmatrix}$$

# Pole Region Constraints

- ▶ the LMI represents a necessary and sufficient condition for the matrix  $A$  to have all eigenvalues in the left half plane.
- ▶ one can express an arbitrary region  $\mathcal{D}$  in the complex plane in terms of two matrix  $L = L^T$  and  $M$  as the set of all complex numbers that satisfy and LMI constraint.

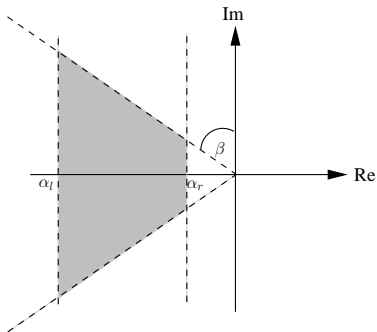
$$\mathcal{D} = \{s \in \mathbb{C} : L + Ms + M^T \bar{s} < 0\}$$

where  $\bar{s}$  denotes the complex conjugate of  $s$ .

- ▶ Such a region is called an **LMI region**.

# Pole Region Constraints

Example: poles region constraint



From  $\operatorname{Re} s < \alpha_r$ , we have

$$\frac{s + \bar{s}}{2} < \alpha_r$$
$$s + \bar{s} - 2\alpha_r < 0.$$

# Pole Region Constraints

Example: poles region constraint

and  $\text{Im } s > \alpha_l$

$$\frac{s + \bar{s}}{2} > \alpha_l$$
$$-s - \bar{s} + 2\alpha_l < 0.$$

Thus

$$L_v = \begin{bmatrix} 2\alpha_l & 0 \\ 0 & -2\alpha_r \end{bmatrix}, \quad M_v = \begin{bmatrix} -1 & 0 \\ 0 & 1 \end{bmatrix}$$

For the conic sector, a complex number  $s = x + jy$  lies in the conic sector if and only if

$$\left| \frac{x}{y} \right| < \tan \beta = \frac{\sin \beta}{\cos \beta} \quad \text{and} \quad x \cos \beta < 0.$$

Rewrite the above conditions in the form

$$\frac{x^2}{y^2} < \frac{\sin^2 \beta}{\cos^2 \beta}$$

# Pole Region Constraints

Example: poles region constraint

we get

$$\begin{aligned}x^2 \cos^2 \beta - y^2 \sin^2 \beta &< 0 \\x \cos \beta - \frac{y^2 \sin^2 \beta}{x \cos \beta} &< 0\end{aligned}$$

By Schur's complement we have

$$M_c s + M_c^* \bar{s} = \begin{bmatrix} 2x \cos \beta & 2jy \sin \beta \\ -2jy \sin \beta & 2x \cos \beta \end{bmatrix} < 0$$

Thus

$$L_c = \begin{bmatrix} 0 & 0 \\ 0 & 0 \end{bmatrix}, \quad M_c = \begin{bmatrix} \cos \beta & -\sin \beta \\ \sin \beta & \cos \beta \end{bmatrix}$$

The two constraints can be combined as

$$L = \begin{bmatrix} L_c & 0 \\ 0 & L_v \end{bmatrix}, \quad M = \begin{bmatrix} M_c & 0 \\ 0 & M_v \end{bmatrix}$$



# Constraints on the $\mathcal{H}_\infty$ Norm

Consider the system with transfer function  $T(s)$  as state space realization

$$\begin{aligned}\dot{x}(t) &= Ax(t) + Bw(t), & x(0) &= 0 \\ z(t) &= Cx(t) + Dw(t)\end{aligned}$$

Assuming that  $T(s)$  is stable, the  $\mathcal{H}_\infty$  norm of the system is

$$\|T\|_\infty^2 = \max_{w \neq 0} \frac{\int_0^\infty z^T(t)z(t)dt}{\int_0^\infty w^T(t)w(t)dt}, \quad x(0) = 0.$$

It follows that  $\|T\|_\infty < \gamma$  is equivalent to

$$\int_0^\infty (z^T(t)z(t) - \gamma^2 w^T(t)w(t))dt < 0$$

Holding true for all square integrable, non-zero  $w(t)$ .

# Constraints on the $\mathcal{H}_\infty$ Norm

Introduce a Lyapunov function  $V(x) = x^T P x$  with  $P = P^T > 0$ . Since  $x(0) = x(\infty) = 0$ , the constraint  $\|T\|_\infty < \gamma$  is enforced by the existence of a matrix  $P = P^T > 0$  such that

$$\frac{dV(x)}{dt} + \frac{1}{\gamma} z^T(t) z(t) - \gamma w^T(t) w(t) < 0$$

for all  $x(t), w(t)$ ; to turn into a LMI, substitute

$$\frac{dV(x)}{dt} = x^T (A^T P + P A) x + x^T P B w + w^T B^T P x, \quad z = C x + D w$$

To obtain

$$\begin{bmatrix} x \\ w \end{bmatrix}^T \begin{bmatrix} A^T P + P A + \frac{1}{\gamma} C^T C & P B + \frac{1}{\gamma} C^T D \\ B^T P + \frac{1}{\gamma} D^T C & -\gamma I + \frac{1}{\gamma} D^T D \end{bmatrix} \begin{bmatrix} x \\ w \end{bmatrix} < 0$$

# Constraints on the $\mathcal{H}_\infty$ Norm

For  $\|T\|_\infty < \gamma$  the above must hold for all  $x$  and  $w$ , thus the block matrix must be negative definite. The condition can be rewritten as

$$\begin{bmatrix} A^T P + PA & PB \\ B^T P & -\gamma I \end{bmatrix} + \frac{1}{\gamma} \begin{bmatrix} C^T \\ D^T \end{bmatrix} \begin{bmatrix} C & D \end{bmatrix} < 0$$

By Schur complement, we have

## Theorem (Bound real lemma)

$\|T\|_\infty < \gamma$  if and only if there exists a positive definite, symmetric matrix  $P$  that satisfies the linear matrix inequality

$$\begin{bmatrix} A^T P + PA & PB & C^T \\ B^T P & -\gamma I & D^T \\ C & D & -\gamma I \end{bmatrix} < 0$$

# Constraints on the $\mathcal{H}_\infty$ Norm

Using the congruence transformation and  $Q = P^{-1}$ ,

$$\begin{bmatrix} Q & 0 & 0 \\ 0 & I & 0 \\ 0 & 0 & I \end{bmatrix} \begin{bmatrix} A^T P + PA & PB & C^T \\ B^T P & -\gamma I & D^T \\ C & D & -\gamma I \end{bmatrix} \begin{bmatrix} Q & 0 & 0 \\ 0 & I & 0 \\ 0 & 0 & I \end{bmatrix} < 0.$$

An equivalent form

$$\begin{bmatrix} QA^T + AQ & B & QC^T \\ B^T & -\gamma I & D^T \\ CQ & D & -\gamma I \end{bmatrix} < 0$$

# Constraints on the $\mathcal{H}_\infty$ Norm

Using `cvx`

```
sys = rss(3,3);
A = sys.a; B = sys.b; C = sys.c; D = sys.d;
n = size(A,1); nu = size(B,2); ny = size(D,1);

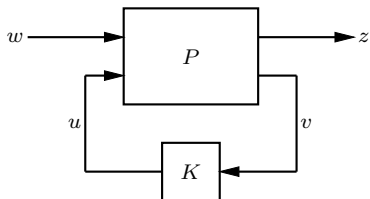
cvx_begin sdp
    variable P(n,n) symmetric
    variable gamma;

    minimize gamma;
    subject to
        P > 0;
        [A'*P + P*A, P*B, C';
         B'*P , -gamma*eye(nu), D';
         C, D, -gamma*eye(ny)] < 0;
cvx_end

display(P);
```

# Controller Design Using LMIs

## Generalized Plant



The generalized plant  $P(s)$  has a state space realization

$$\dot{x}(t) = Ax(t) + B_w w(t) + B_u u(t)$$

$$z(t) = C_z x(t) + D_{zw} w(t) + D_{zu} u(t)$$

$$v(t) = C_v x(t) + D_{vw} w(t)$$

# Controller Design Using LMIs

## $\mathcal{H}_\infty$ State Feedback

State feedback  $u = Fx$  yields the closed-loop system

$$\dot{x}(t) = (A + B_u F)x(t) + B_w w(t)$$

$$z(t) = (C_z + D_{zu} F)x(t) + D_{zw} w(t)$$

Replacing the system matrices in Bounded real lemma by the closed-loop matrices and using the variable transformation  $Y = FQ$  leads to the following result: a necessary and sufficient condition for a state feedback controller to achieve a  $\mathcal{H}_\infty$ -norm less than  $\gamma$  is the existence of matrices  $P = P^T > 0$  and  $Y$  that satisfy

$$\begin{bmatrix} QA^T + AQ + Y^T B_u^T + B_u Y & B_w & QC_z^T + Y^T D_{zu}^T \\ B_w^T & -\gamma I & D_{zw}^T \\ C_z Q + D_{zu} Y & D_{zw} & -\gamma I \end{bmatrix} < 0, \quad F = YQ^{-1}$$

# Controller Design Using LMIs

$\mathcal{H}_\infty$  State Feedback

$$G(s) = \left[ \begin{array}{c|cc} A & B_w & B_u \\ \hline C_z & D_{zw} & D_{zu} \\ I & 0 & 0 \end{array} \right] \quad \text{with } (A, B_u) \text{ assumed to be stabilizable}$$

minimize  $\gamma$

subject to

$$Q = Q^T > 0$$

$$\begin{bmatrix} QA^T + AQ + Y^T B_u^T + B_u Y & B_w & QC_z^T + Y^T D_{zu}^T \\ B_w^T & -\gamma I & D_{zw}^T \\ C_z Q + D_{zu} Y & D_{zw} & -\gamma I \end{bmatrix} < 0$$

If this has a solution then

$$F = YQ^{-1}$$



# $\mathcal{H}_\infty$ State Feedback Design

Using `cvx`

```
G = ss(A, [Bw, Bu], [Cz; Cv], [Dzw, Dzu; Dvw, zeros(nv,nu)]);
```

```
cvx_begin sdp
    variable Q(n,n) symmetric;
    variable Y(nu,n);
    variable gamma;

    minimize gamma;
    subject to
        Q > 0;
        [Q*A' + A*Q + Bu*Y + Y'*Bu',    Bw,    Q*Cz' + Y'*Dzu';
         Bw',    -gamma*eye(nu,nu),    Dzw'];
        Cz*Q + Dzu*Y,    Dzw,    -gamma*eye(nv,nv)] < 0;
cvx_end

F = Y*inv(Q);

% check closed-loop poles
Aclp = A + Bu*F;
disp(eig(Aclp));
```

# Controller Design Using LMIs

## Controller dynamics

The controller dynamics are represented by a state space model

$$\begin{aligned}\dot{\zeta}(t) &= A_K \zeta(t) + B_K v(t) \\ u(t) &= C_K \zeta(t) + D_K v(t)\end{aligned}$$

The state space realization of the closed loop-system is

$$\begin{aligned}\dot{x}_c(t) &= A_c x_c(t) + B_c w(t) \\ z(t) &= C_c x_c(t) + D_c w(t)\end{aligned}$$

where

$$\begin{aligned}A_c &= \begin{bmatrix} A + B_u D_K C_v & B_u C_K \\ B_K C_v & A_K \end{bmatrix}, & B_c &= \begin{bmatrix} B_w + B_u D_K D_{vw} \\ B_K D_{vw} \end{bmatrix}, \\ C_c &= \begin{bmatrix} C_z + D_{zu} D_K C_v & D_{zu} C_K \end{bmatrix}, & D_c &= D_{zw} + D_{zu} D_K D_{vw}\end{aligned}$$

# Controller Design Using LMIs

## $\mathcal{H}_\infty$ Output Feedback

$$P(s) = \left[ \begin{array}{c|cc} A & B_w & B_u \\ \hline C_z & D_{zw} & D_{zu} \\ C_v & D_{vw} & 0 \end{array} \right]$$

with  $(A, B_u)$  assumed to be stabilizable  
and  $(C_v, A)$  assumed to be detectable

for output feedback (assume  $D_K = 0$ ):

$$u = K(s)y = \left[ \begin{array}{c|c} A_K & B_K \\ \hline C_K & 0 \end{array} \right] y$$

$$G(s) = \mathcal{F}_l(P(s), K(s)) = \left[ \begin{array}{cc|c} A & B_u C_K & B_w \\ \hline B_K C_v & A_K & B_K D_{vw} \\ \hline C_z & D_{zu} C_K & D_{zw} \end{array} \right]$$

$$G(s) = \left[ \begin{array}{c|c} A_c & B_c \\ \hline C_c & D_c \end{array} \right]$$

# Controller Design Using LMIs

## $\mathcal{H}_\infty$ Output Feedback

The LMI condition is :

$$\begin{bmatrix} A_c^T P + P A_c & P B_c & C_c^T \\ B_c^T P & -\gamma I & D_c^T \\ C_c & D_c & -\gamma I \end{bmatrix} < 0$$

Partition  $P$  as:

$$P = \begin{bmatrix} X & R \\ R^T & * \end{bmatrix} \text{ and } P^{-1} = \begin{bmatrix} Y & S \\ S^T & * \end{bmatrix}$$

Define an inertia-preserving transform via:

$$P T_Y = T_X \text{ where } T_Y = \begin{bmatrix} Y & I \\ S^T & 0 \end{bmatrix}, \quad T_X = \begin{bmatrix} I & X \\ 0 & R^T \end{bmatrix}$$

# Controller Design Using LMIs

## $\mathcal{H}_\infty$ Output Feedback

The matrices  $T_X$  and  $T_Y$  can be used to transform the nonlinear constraint into a linear one. This transformation is based on the fact that

$$T_Y^T P A_c T_Y = T_X^T A_c T_Y = \begin{bmatrix} AY + B_u \tilde{C}_K & A \\ \tilde{A}_K & XA + \tilde{B}_K C_v \end{bmatrix}$$
$$T_Y^T P B_c = \begin{bmatrix} B_w \\ XB_w + \tilde{B}_K D_{vw} \end{bmatrix}, \quad C_c T_Y = \begin{bmatrix} C_z Y + D_{zu} \tilde{C}_K & C_z \end{bmatrix}$$

where

$$\tilde{A}_K = RA_K S^T + RB_K C_v Y + XB_u C_K S^T + XAY$$

$$\tilde{B}_K = RB_K$$

$$\tilde{C}_K = C_K S^T$$

$$\tilde{D}_K = D_K$$

$$T_Y^T P T_Y = \begin{bmatrix} Y & I \\ I & X \end{bmatrix}$$

# Controller Design Using LMIs

## $\mathcal{H}_\infty$ Output Feedback

The LMI condition is :

$$\begin{bmatrix} A_c^T P + P A_c & P B_c & C_c^T \\ B_c^T P & -\gamma I & D_c^T \\ C_c & D_c & -\gamma I \end{bmatrix} < 0, \text{ and } P > 0.$$

$$\begin{bmatrix} T_Y^T & 0 & 0 \\ 0 & I & 0 \\ 0 & 0 & I \end{bmatrix} \begin{bmatrix} A_c^T P + P A_c & P B_c & C_c^T \\ B_c^T P & -\gamma I & D_c^T \\ C_c & D_c & -\gamma I \end{bmatrix} \begin{bmatrix} T_Y & 0 & 0 \\ 0 & I & 0 \\ 0 & 0 & I \end{bmatrix} =$$

$$\begin{bmatrix} AY + YA^T + B_u \tilde{C}_K + \tilde{C}_K^T B_u^T & \tilde{A}_K^T + A & B_w & Y C_z^T + \tilde{C}_K^T D_{zu}^T \\ * & A^T X + X A + \tilde{B}_K C_v + C_v^T \tilde{B}_K^T & X B_w + \tilde{B}_K D_{vw} & C_z^T \\ * & * & -\gamma I & D_{zw}^T \\ * & * & * & -\gamma I \end{bmatrix}$$

# Controller Design Using LMIs

## $\mathcal{H}_\infty$ Output Feedback

$$\begin{array}{ll} \text{minimize} & \gamma \\ \text{subject to :} & \begin{bmatrix} Y & I \\ I & X \end{bmatrix} > 0 \end{array}$$

$$\begin{bmatrix} AY + YA^T + B_u \tilde{C}_K + \tilde{C}_K^T B_u^T & \tilde{A}_K^T + A & B_w & Y C_z^T + \tilde{C}_K^T D_{zu}^T \\ * & A^T X + XA + \tilde{B}_K C_v + C_v^T \tilde{B}_K^T & X B_w + \tilde{B}_K D_{vw} & C_z^T \\ * & * & -\gamma I & D_{zw}^T \\ * & * & * & -\gamma I \end{bmatrix} < 0$$

If this has a solution  $\gamma, X, Y, \tilde{A}_K, \tilde{B}_K$  and  $\tilde{C}_K$  then

$$PP^{-1} = I \quad \implies \quad RS^T = I - YX \quad (\text{solve for } R \text{ and } S)$$

Solve for  $A_K, B_K$  and  $C_K$  from:

$$\begin{aligned} \tilde{A}_K &= RA_K S^T + RB_K C_v Y + X B_u C_K S^T + XAY \\ \tilde{B}_K &= RB_K \\ \tilde{C}_K &= C_K S^T \end{aligned}$$

# $\mathcal{H}_\infty$ Output Feedback Design

Using `cvx`

```
G = ss(A, [Bw, Bu], [Cz; Cv], [Dzw, Dzu; Dvw, zeros(nv,nu)]);
cvx_begin sdp
    variable X(n,n) symmetric;
    variable Y(n,n) symmetric;
    variable Ah(n,n); % Ah is a tilde A
    variable Bh(n,n); % Bh is a tilde B
    variable Ch(nu,n); % Ch is a tilde C
    variable gamma;

    minimize gamma;
    subject to

        [Y, eye(n,n);
         eye(n,n), X] > 0;

        [A*Y + Bu*Ch + Y*A' + Ch'*Bu', A+Ah', Bw, Y*Cz' + Ch'*Dzu';
         A'+Ah, X+A + A'*Y + Bh*Cv + Cv'*Bh', X*Bw + Bh*Dvw, Cz';
         Bw', Bw'*X + Dvw'*Bh', -gamma*eye(nw,nw), Dzw';
         Cz*Y + Dzu*Ch, Cz, Dzw, -gamma*eye(nz,nz)] < 0;
cvx_end
```



# $\mathcal{H}_\infty$ Output Feedback Design

Using `cvx`

```
% Reconstruct the controller by inverting the linearizing transform.
Mnt = eye(n,n) - X*Y;           % not symmetric
[Umn, Smn, Vmn] = svd(Mnt);

sSmn = sqrt(diag(Smn))          % take the square roots
isSmn = 1./sSmn;               % their inverse

M = Umn*diag(sSmn);           N = Vmn*diag(sSmn);
% Calculate inverses
iM = diag(isSmn)*Umn';       iN = diag(isSmn)*Vmn';
% Real use you should check whether the inverse is succeeded or not

DK = zeros(nu,nv);
BK = iN*Bh;
CK = Ch*(iM)';
AK = iN*(Ah - Bh*Cv*X - Y*Bu*Ch - Y*A*X)*(iM)';

K = ss(AK,BK,CK,DK);
```

- 1 Herbert Werner "Lecture note on *Optimal and Robust Control*", 2012
- 2 Roy Smith "Lecture note on *Robust Control & Convex Optimization*", 2012